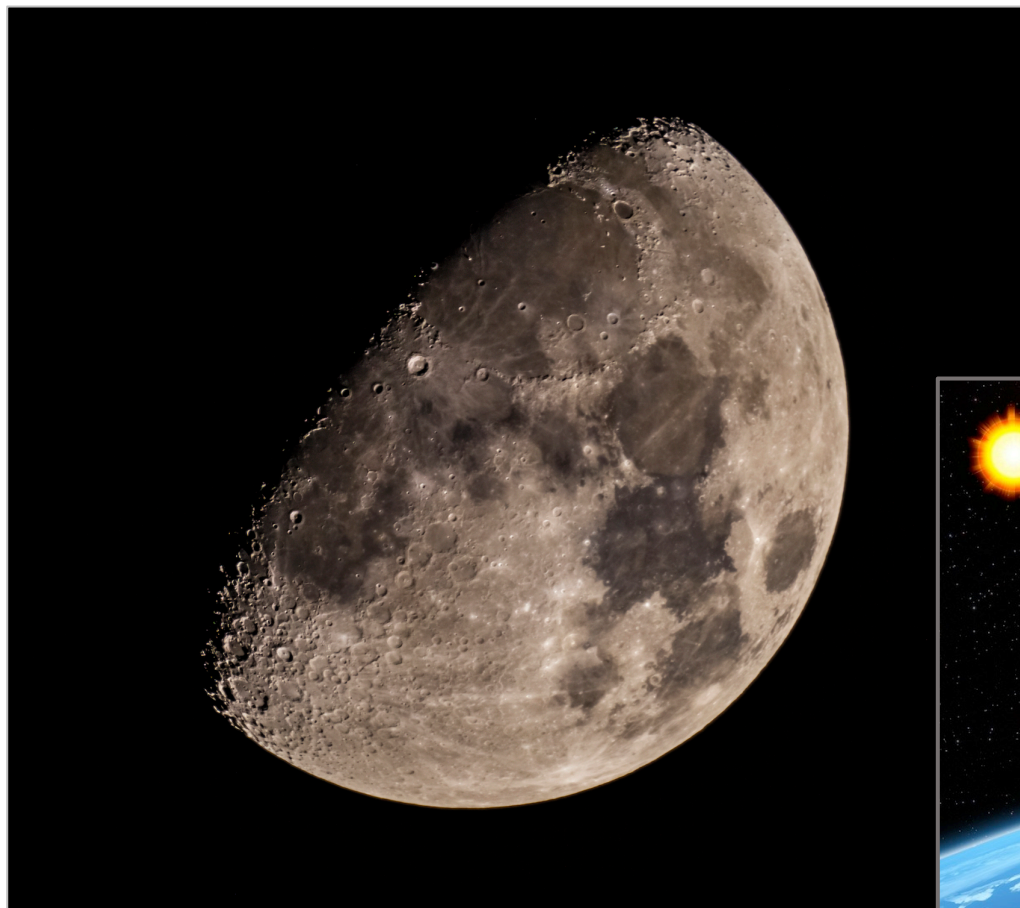


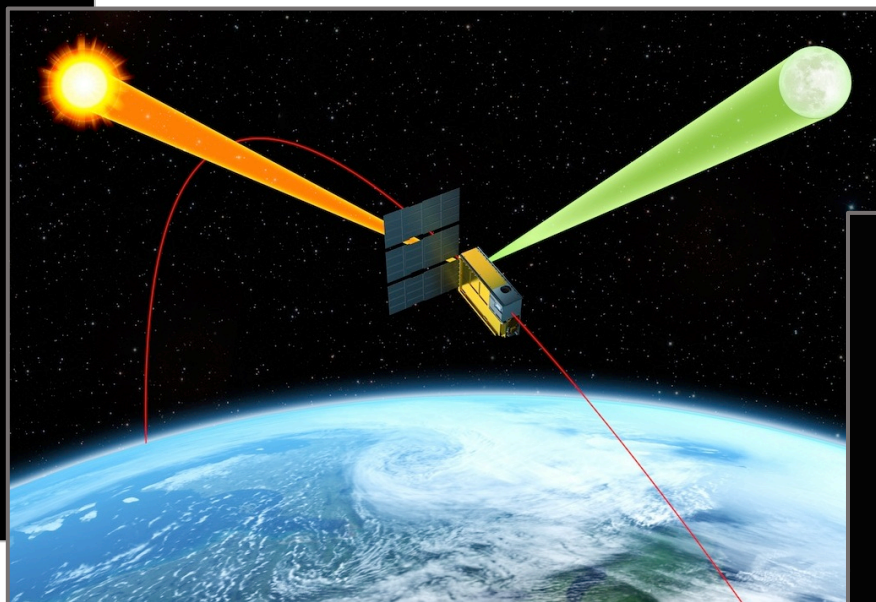
ARCSTONE

T. Stone, C. Lukashin, et al., "Redeveloping the Lunar Reflectance as a High-accuracy Absolute Reference for On-orbit Radiometric Calibration," input to NRC Decadal Survey, 2017.

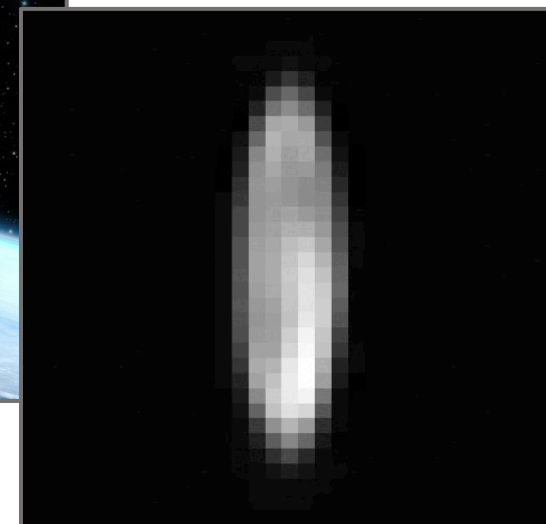


Lunar Surface Reflectance: Invariable $< 10^{-8}$

**ARCSTONE:
Calibration of Lunar Spectral Reflectance
from Space**



Sensor Calibration



Accurate measurements of Lunar Irradiance from Space
Lunar Irradiance Model



ARCSTONE Team

Principal Investigator: Constantine Lukashin¹

Lead Engineer: Trevor Jackson¹ (presenting)

**Team: T. Stone², G. Kopp³, R. Swanson⁴, C. Buleri⁵, T. Christianson⁵
M. Cooney¹, H. Courrier⁴, W. Davis¹, M. Kehoe⁴, T. Nguyen¹
G. Rutherford¹, N. Ryan¹, P. Smith³, M. Stebbins⁴, C. Young¹**

1 – NASA Langley Research Center, Hampton, VA

2 – USGS, Flagstaff, AZ

3 – LASP Colorado U, Boulder, CO

4 – Resonon Inc., Bozeman, MT

5 – Quartus Engineering, San Diego, CA

PARTNERS:

USGS

LASP

Resonon

**Quartus
Engineering**

**Blue Canyon
Technologies**

NIST

GSFC

ARCSTONE Moon as Solar Diffuser For Instrument Calibration

Current EOS designed on overlapping observations – Cannot handle data gaps !
(CERES, MODIS/VIIRS, Landsat, PACE/SeaWIFS, etc.)

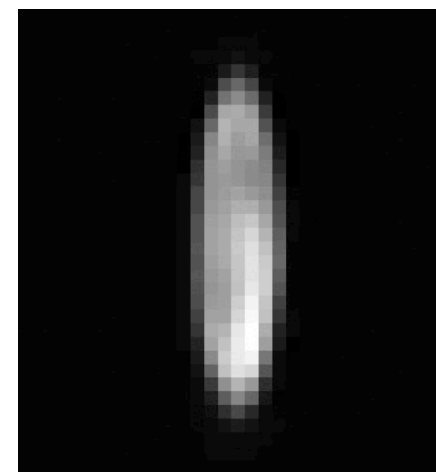


Reflectance of Lunar surface stable to $< 10^{-8}$ / year
Calibration reference: Lunar irradiance (entire disk)

Remote Sensing Calibration Need:

Absolute accurate spectral irradiance
for all lunar phase and libration states !

Current ROLO Accuracy: estimated 5 – 10%



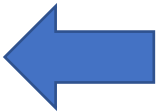
Lunar image by SeaWIFS

SeaWiFS gain stability 0.13% ($k=1$) over 12 years -- achieved with Lunar calibration

ARCS^{ONE}

Applications of the Lunar Calibration Approach

Team	Satellite	Sensor	G/L	Dates	Number of obs	Phase angle range (°)
CMA	FY-3C	MERSI	LEO	2013-2014	9	[43 57]
CMA	FY-2D	VISSR	GEO	2007-2014		
CMA	FY-2E	VISSR	GEO	2010-2014		
CMA	FY-2F	VISSR	GEO	2012-2014		
JMA	MTSAT-2	IMAGER	GEO	2010-2013	62	[-138,147]
JMA	GMS5	VISSR	GEO	1995-2003	50	[-94,96]
JMA	Himawari-8	AHI	GEO	2014-	-	
EUMETSAT	MSG1	SEVIRI	GEO	2003-2014	380/43	[-150,152]
EUMETSAT	MSG2	SEVIRI	GEO	2006-2014	312/54	[-147,150]
EUMETSAT	MSG3	SEVIRI	GEO	2013-2014	45/7	[-144,143]
EUMETSAT	MET7	MVIRI	GEO	1998-2014	128	[-147,144]
CNES	Pleiades-1A	PHR	LEO	2012	10	[+/-40]
CNES	Pleiades-1B	PHR	LEO	2013-2014	10	[+/-40]
NASA-MODIS	Terra	MODIS	LEO	2000-2014	136	[54,56]
NASA-MODIS	Aqua	MODIS	LEO	2002-2014	117	[-54,-56]
NASA-VIIRS	NPP	VIIRS	LEO	2012-2014	20	[50,52]
NASA-OBPG	SeaStar	SeaWiFS	LEO	1997-2010	204	(<10, [27-66])
NASA/USGS	Landsat-8	OLI	LEO	2013-2014	3	[-7]
NASA	OCO-2	OCO	LEO	2014		
NOAA-STAR	NPP	VIIRS	LEO	2011-2014	19	[-52,-50]
NOAA	GOES-10	IMAGER	GEO	1998-2006	33	[-66, 81]
NOAA	GOES-11	IMAGER	GEO	2006-2007	10	[-62, 57]
NOAA	GOES-12	IMAGER	GEO	2003-2010	49	[-83, 66]
NOAA	GOES-13	IMAGER	GEO	2006	11	
NOAA	GOES-15	IMAGER	GEO	2012-2013	28	[-52, 69]
VITO	Proba-V	VGT-P	LEO	2013-2014	25	[-7]
KMA	COMS	MI	GEO	2010-2014	60	
AIST	Terra	ASTER	LEO	1999-2014	1	-27.7
ISRO	OceanSat2	OCM-2	LEO	2009-2014	2	
ISRO	INSAT-3D	IMAGER	GEO	2013-2014	2	

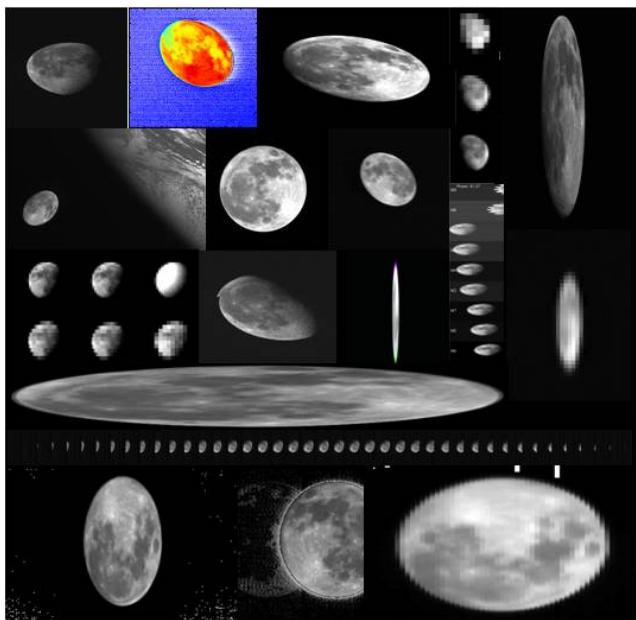


Instruments with lunar observation capabilities participating in the GSICS GIRO program: All satellite operators !

From GSICS Lunar Calibration Workshop, December 2014, EUMETSAT.

Not included by GSICS: commercial land imager constellations (e.g. Planet)

Images of the Moon from spaceborne sensors



ARCSTONE

Mission Concept

Concept of Operations and Data Products:

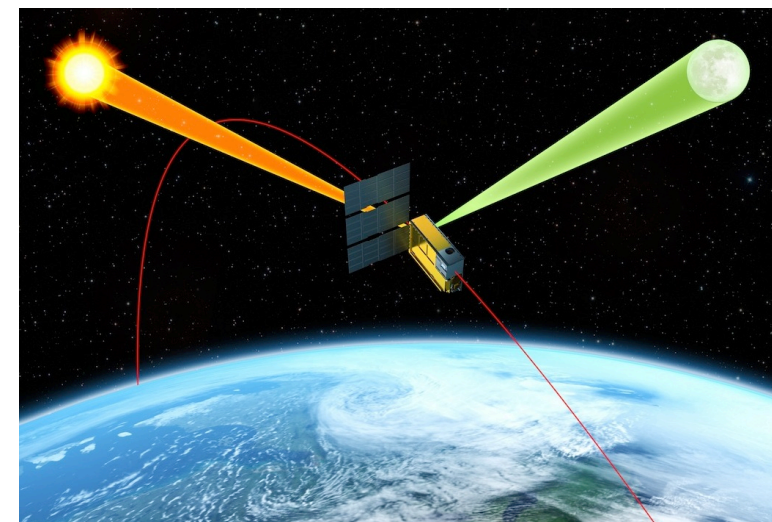
- Data to collect: Lunar spectral irradiance every 12 hours, 10 minutes
- Data to collect: Solar spectral irradiance for calibration (at least weekly)
- Combined uncertainty $< 0.5\%$ ($k=1$)
- Grating spectrometer with single field-of-view about 0.6° (no scanning !)
- Spectral range from 350 nm to 2300 nm, spectral sampling at 4 nm
- Sun synchronous orbit at 500 – 600 km altitude

1 year: Improvement of current Lunar Calibration Model (factor > 2);

3+ years: New Lunar Irradiance Model, improved accuracy level (factor > 10).

Key Technologies to Enable the Concept:

- Approach to orbital calibration via referencing Sun (TSIS measurements):
Demonstration of lunar and solar measurements with *the same optical path using integration time to reduce solar signal -- Major Innovation !*
- Pointing ability of small spacecraft now permits obtaining required measurements *with instrument integrated into spacecraft.*



6U CubeSat Spacecraft Bus:
courtesy of Blue Canyon Technologies



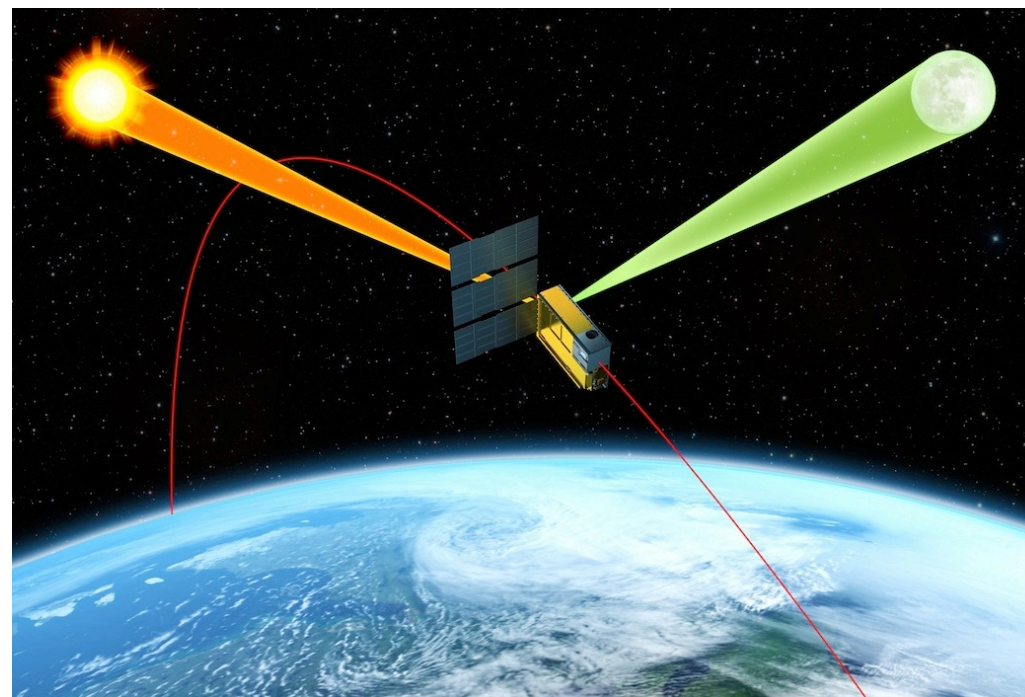
CONOPS & Calibration Strategy

Operation Modes:

1. Lunar spectral irradiance observations:
 - Every 12 hours
 - Close to polar locations
 - Multiple measurements within 5 minutes to get required SNR
2. Solar Spectral Irradiance observations (solar calibration):
 - Every TBD days (e.g. daily)
 - This is radiometric calibration to the TSIS reference
3. Dark images:
 - Multiple measurements with closed shutter
 - Before every lunar and solar observations
4. Dark field (to calibrate offsets):
 - Multiple measurements of dark space
 - TBD (e.g. daily)
5. Spectral calibration:
 - On-board spectral calibration (TBD)
 - Occultation measurement of atmosphere (absorption lines)
6. Spacecraft pointing calibration (and other checks):
 - Defined by the BCT for calibration of spacecraft functions
7. Stand by mode:
 - Mode between lunar and solar observations
8. Safe mode
9. On-board data processing mode
10. Data downlink mode

ARCSTONE Calibration Strategy:

- Detailed ground characterization
- Absolute calibration on-orbit referencing SSI established by the TSIS/SIM
- Calibration and measurements with the same optical path !



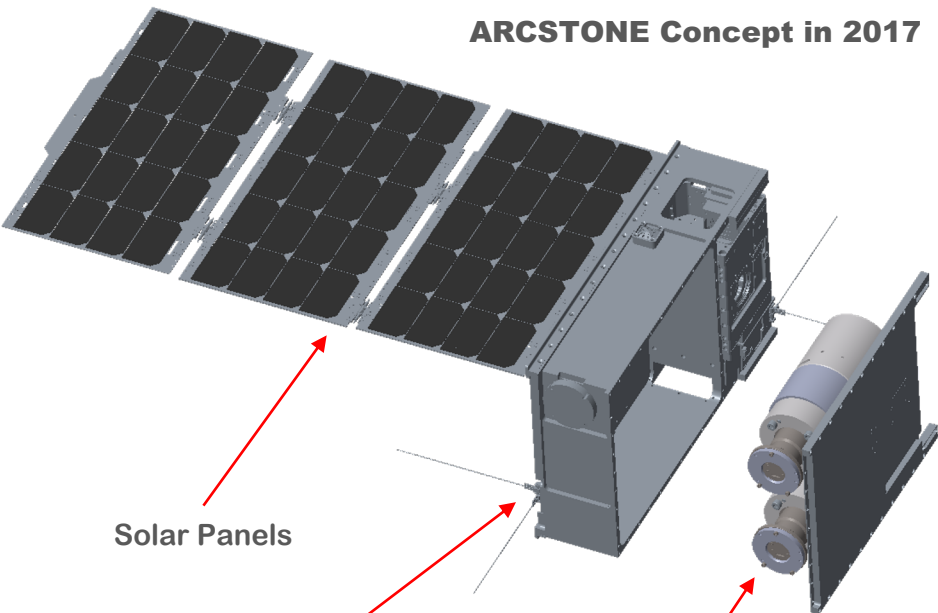
ARCSTONE Absolute Calibration On-orbit

ARCSTONE

Key Mission Performance Parameters

Key Parameters	Threshold Value	Goal Value
Accuracy (reflectance)	1.0% (k=1)	0.5% (k=1)
Stability	< 0.15% (k=1) per decade	< 0.1% (k=1) per decade
Orbit	Sun-synch orbit	Sun-synch orbit
Time on-Orbit	1 year	3 years
Frequency of sampling	24 hours	12 hours
Instrument pointing	< 0.1° combined	< 0.05° combined
Spectral Range	380 nm – 900 nm	350 nm – 2300 nm
Spectral Sampling	8 nm	4 nm

BCT 6U XB6 Spacecraft pointing:
 - Accuracy 0.002° (1-sigma) in 3 axis
 - Stability 1 arc-sec over 1 sec



ARCSTONE Concept in 2017

Solar Panels

ARCSTONE Payload/
Spectrometers:
 - UVVNIR (350 – 880 nm)
 - SWIR (880 – 2300 nm)

Blue Canyon Technologies (BCT)
6U Spacecraft Bus

* Requirements are captured in a Mission Requirements Doc

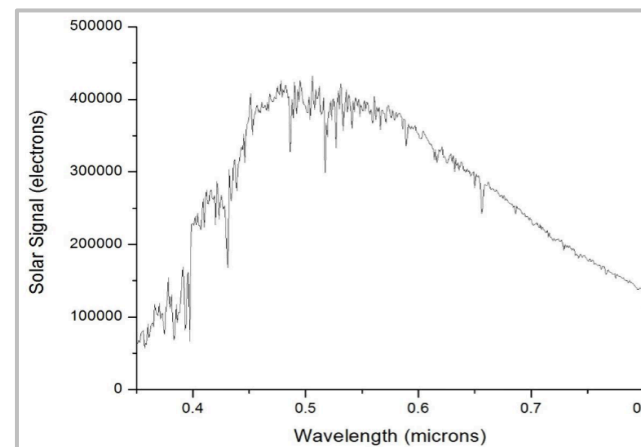
** Threshold Values considered as success criteria

ARCSTONE Project Timeline

Concept brainstorming: down-select on January 9, 2016
 Breadboard design & build by March 2016
 Breadboard tested at NIST in April 2016

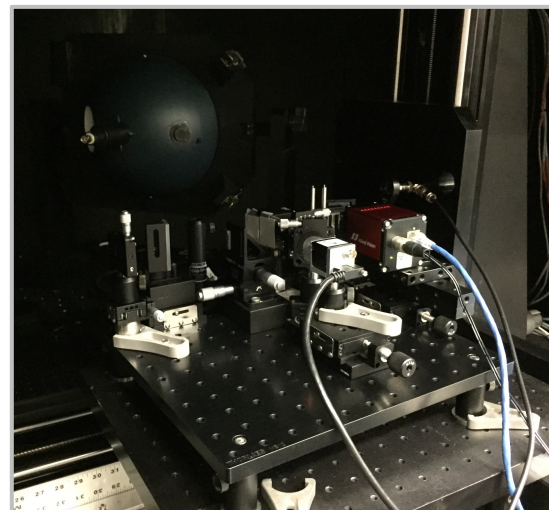
NASA LaRC et al., IIP proposal funded in March, 2017
 Resonon Inc., SBIR Phase-2 funded July, 2017

- Formulation and preliminary design, Summer 2017
- Design and STOP analysis completed by March 2018
- Prep. work for fabrication completed by May 2018
- UVVNIR detector TVAC tested by Fall 2018
- UVVNIR FPA characterized by January 2019
- Fabrication completed by February 2019
- UVVNIR assembled/aligned by March 2019
- **New FSR instrument concept, March 2019**
- SWIR assembly/alignment: on hold
- SWIR IDCA characterization, July 2019
- UVVNIR Characterization: Summer 2019



Solar Spectral Irradiance
 measured by ARCSTONE

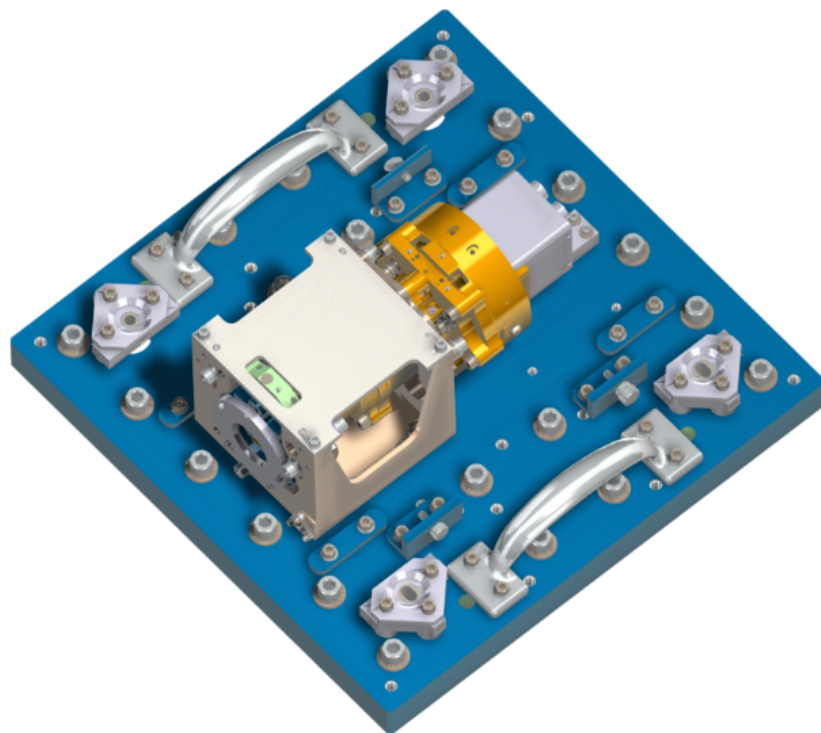
Breadboard at NIST
 SIRCUS Facility, April 2016



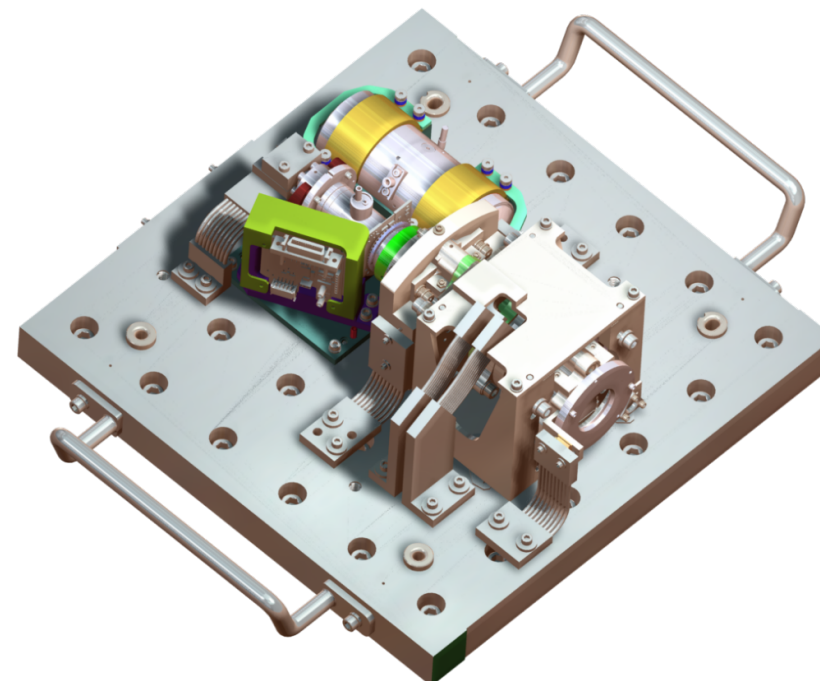
Breadboard at NIST's roof, April 2016

ARCSTONE IIP designed Instruments (EDU)

- Designed for Development Vibe/TVAC but not flight hardware, UVVNIR complete in March 2018
- This is what went into fabrication in April/May 2018, completed in February 2019

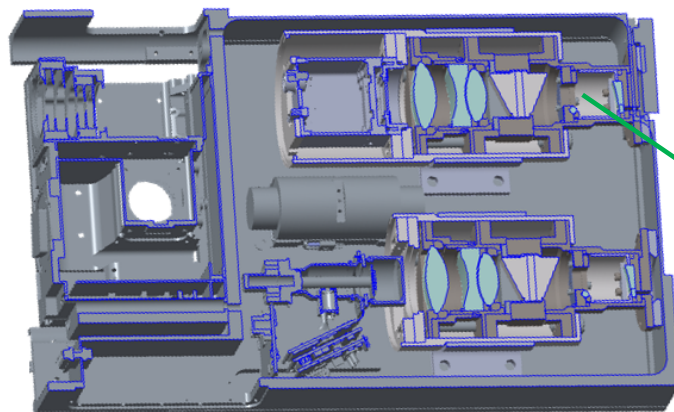


UVVNIR Spectrometer – Ultraviolet Visible Near Infrared
350-880 nm
Transmission Grating Spectrometer
Uncooled FPA and Optic Train



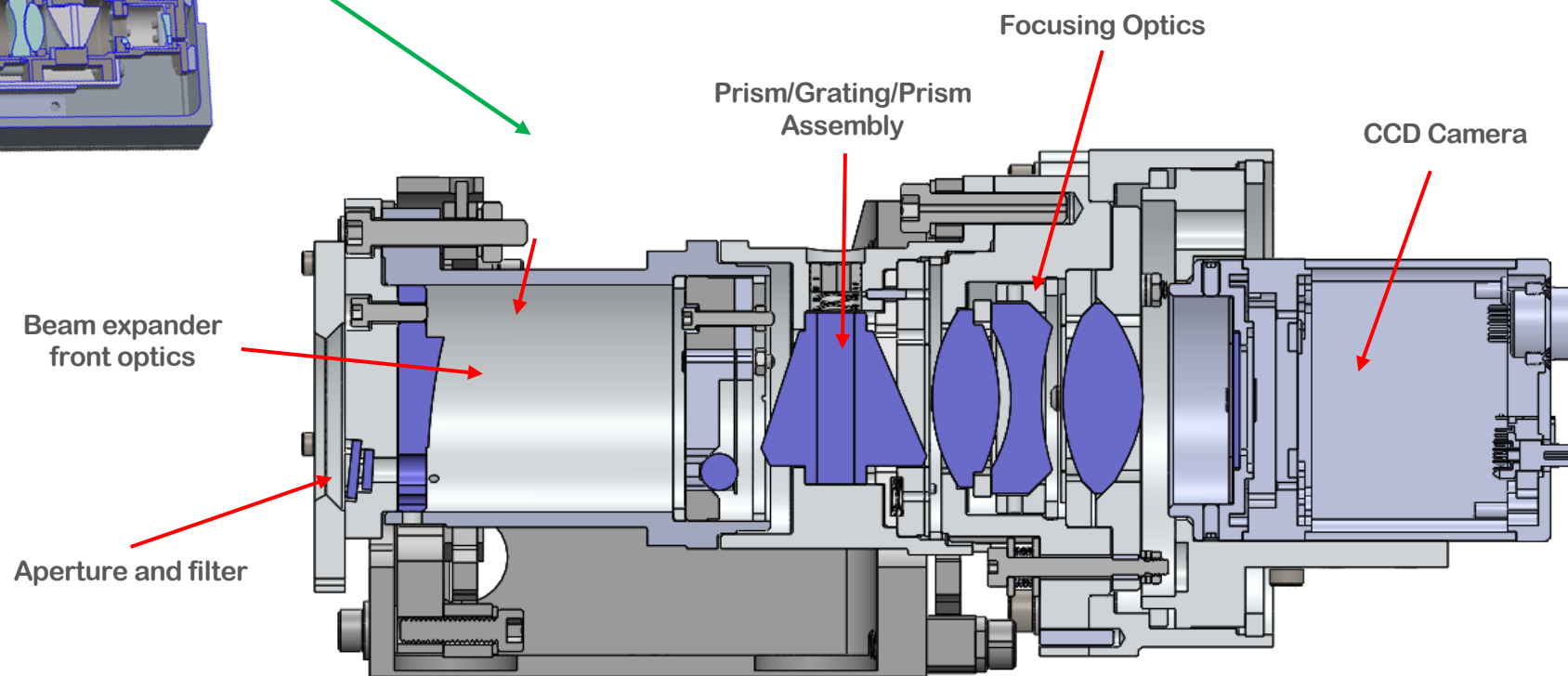
SWIR Spectrometer – Short Wave Infrared
880-2300 nm
Transmission Grating Spectrometer
Cooled FPA and Optic Train

ARCSTONE Observatory and Instrument Design



Cubesat 6U bus:
Courtesy of Blue Canyon Technologies

Preliminary System Design



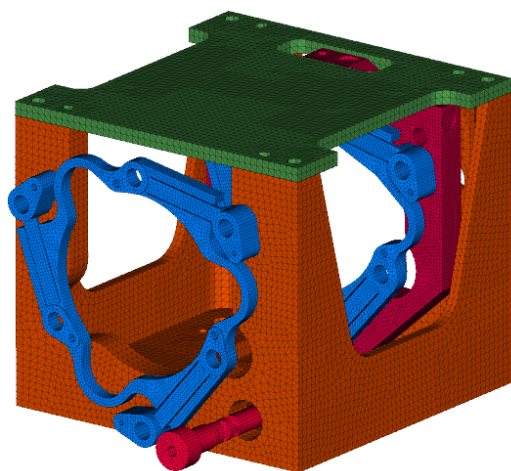
Prototype of ARCSTONE UUVNIR Instrument



STOP Analysis by Quartus Engineering

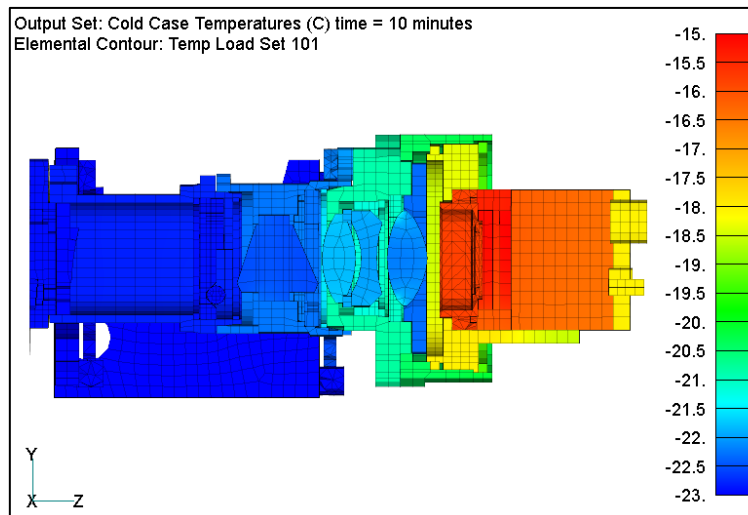
Orbital States from LaRC team →

Thermal Modeling
 Temperature Mapping
 Thermoelastic Modeling
 Nonlinear Analysis Considerations
 Zernike Calculation
 Zemax Modeling
 Thermoelastic Changes of Lenses
 Optical Effects of Thermal Stress
 Mount Modifications of Lenses

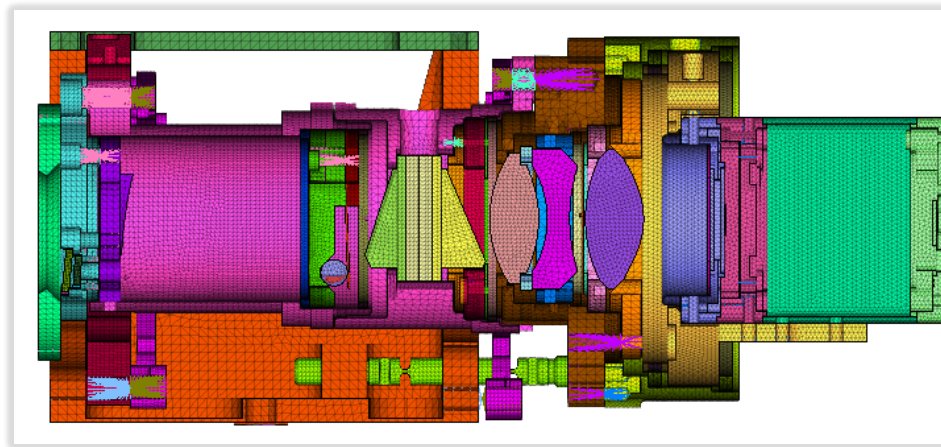


UVVNIR Mount Random Vibration

Current design assumes large thermal operational range in-orbit !



Temperature Mapping Results

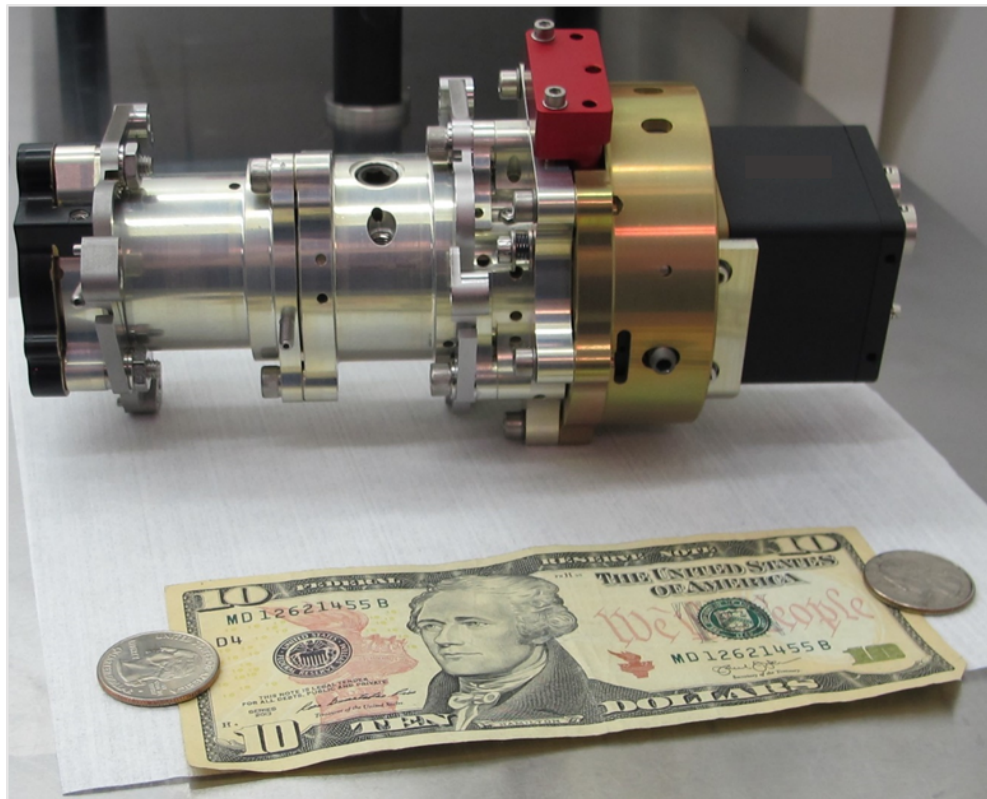


Thermoelastic Results



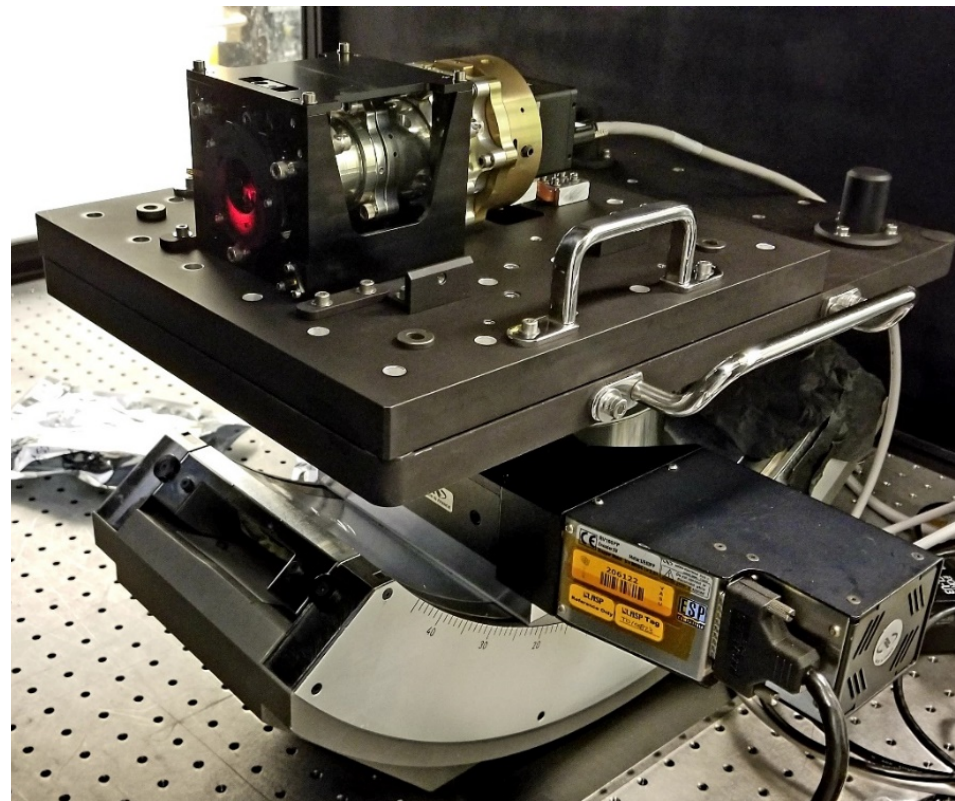
Fabricated & Assembled UVVNIR

Path Forward for UVVNIR Spectrometer: Field campaigns and vibe/TVAC tests



UVVNIR Instrument:

- CCD characterized
- Assembly & Alignment completed



UVVNIR Instrument:

- Undergoing characterization at LASP

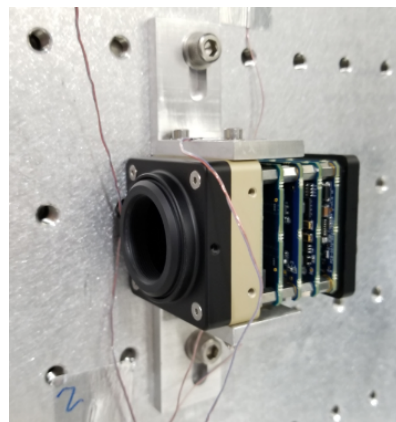
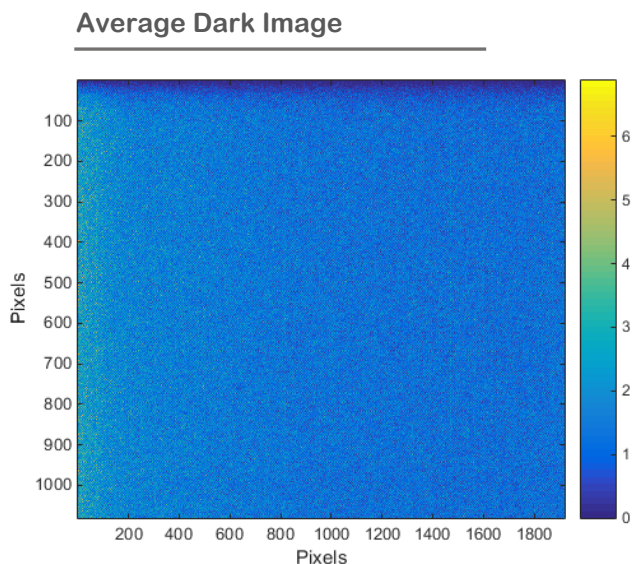
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UVVNIR CCD Camera Characterization

- Sensor is remarkably uniform
 - ✧ No hot pixels
 - ✧ No dead pixels
 - ✧ Only 10's of pixels with $> 5\sigma$ average noise
- Narrow band (20-30 pixels) of pixels with lower fixed-pattern noise offset at top of sensor
- Appears to have ~ 1 DN electronic checkerboard pattern noise (correctable via dark-image subtraction)

UVVNIR CCD Camera Conclusions:

- Camera is consistent to be usable in high-accuracy measurements, with $\sim 0.1\% - 0.3\%$ ($k=1$) uncertainty
 - ✧ *Final uncertainty values will depend on number of pixels used for averaging*
- Dark current is much lower than expected
- There are a number of features that can be overcome through dark-image subtraction and flat-field correction



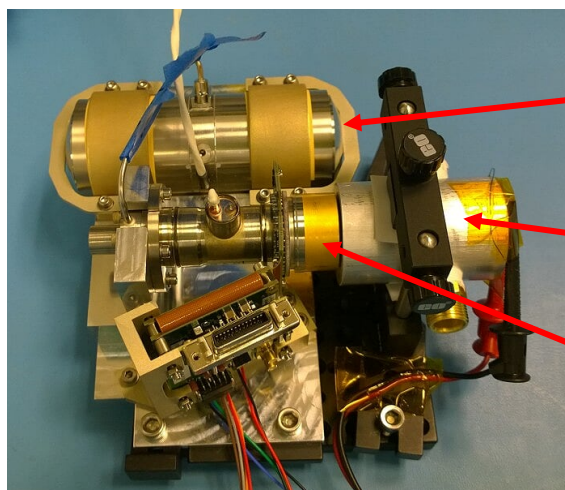
- UUVNIR CCD camera characterized at LASP
- UUVNIR CCD camera TVAC tested at LaRC

Integration time stability and linearity over 10^5 range !

ARCSTONE

SWIR Camera Characterization MCT Integrated Detector Cooler Assembly (IDCA)

- Sensor is uniform
 - ✧ 745 hot/dead pixels
 - ✧ Only 2 pixels with no normal surrounding pixels
- Vertical banding apparent in both dark and light images
 - ✧ Eliminated through dark subtraction



Cryocooler

Cold Source

MCT Detector

Major Credits:

- IDCA selection/acceptance: Mike Cooney (NASA LaRC)
- Mechanical design: Trevor Jackson (NASA LaRC)
- IDCA characterization: Paul Smith (LASP, CU)

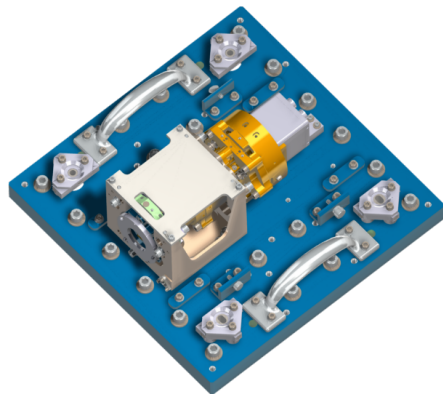
SWIR IDCA Camera Conclusions:

- Camera is expected to be **usable with 0.3% - 0.4% uncertainty depending on signal level:**
 - ✧ *Primary contributor to uncertainty is variation in the offset value between its measurements (repeatability)*
 - ✧ *Offset value variation is a systematic uncertainty that cannot be mitigated through increased averaging, but may be lower during real data collecting operations, e.g. measuring offset before every lunar observation*
- Camera linearity is better than expected
- Final random uncertainty values will depend on number of pixels used for averaging

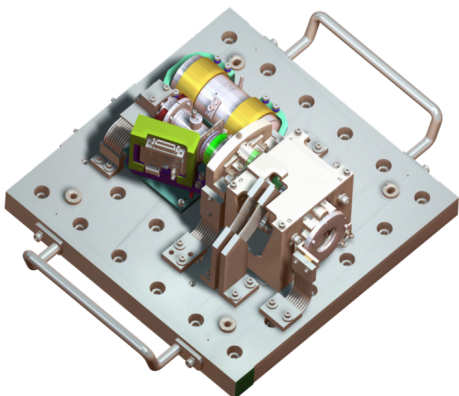
Integration time from 10^{-4} to 3.3 seconds !

ARCSTONE Full Spectral Range (FSR) Instrument

Ultraviolet Visible Near Infrared (UVVNIR)

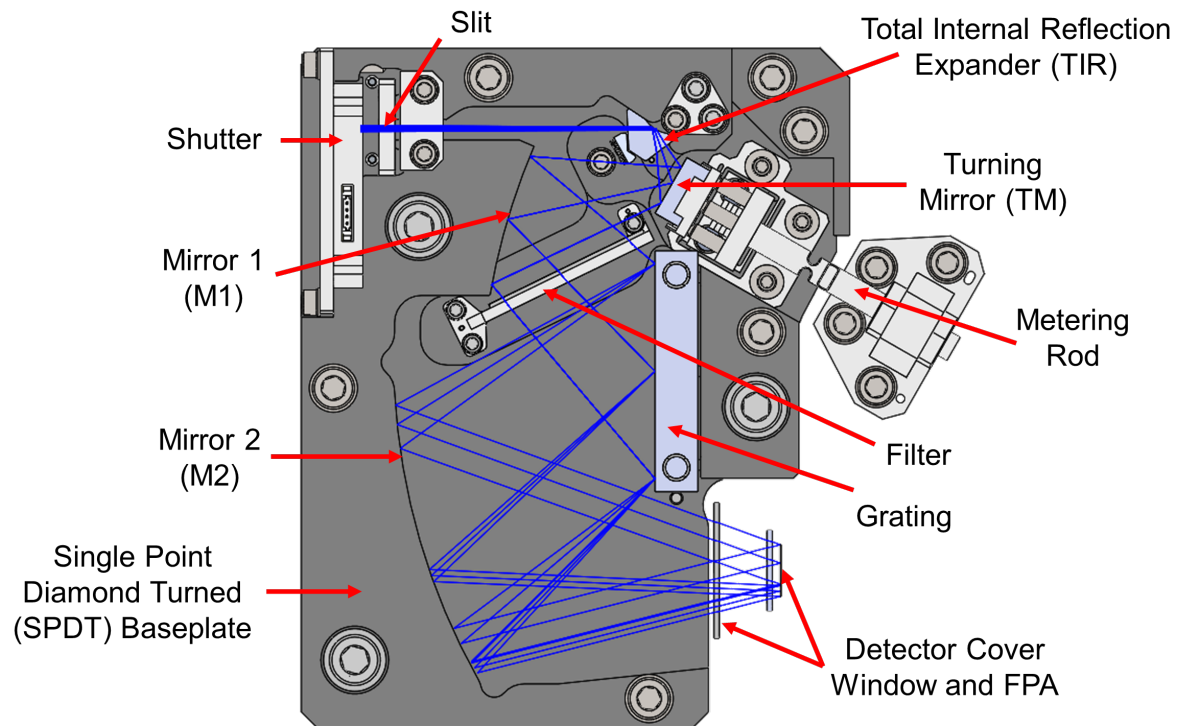


2017 – 2019 IIP work



Short Wave Infrared (SWIR)

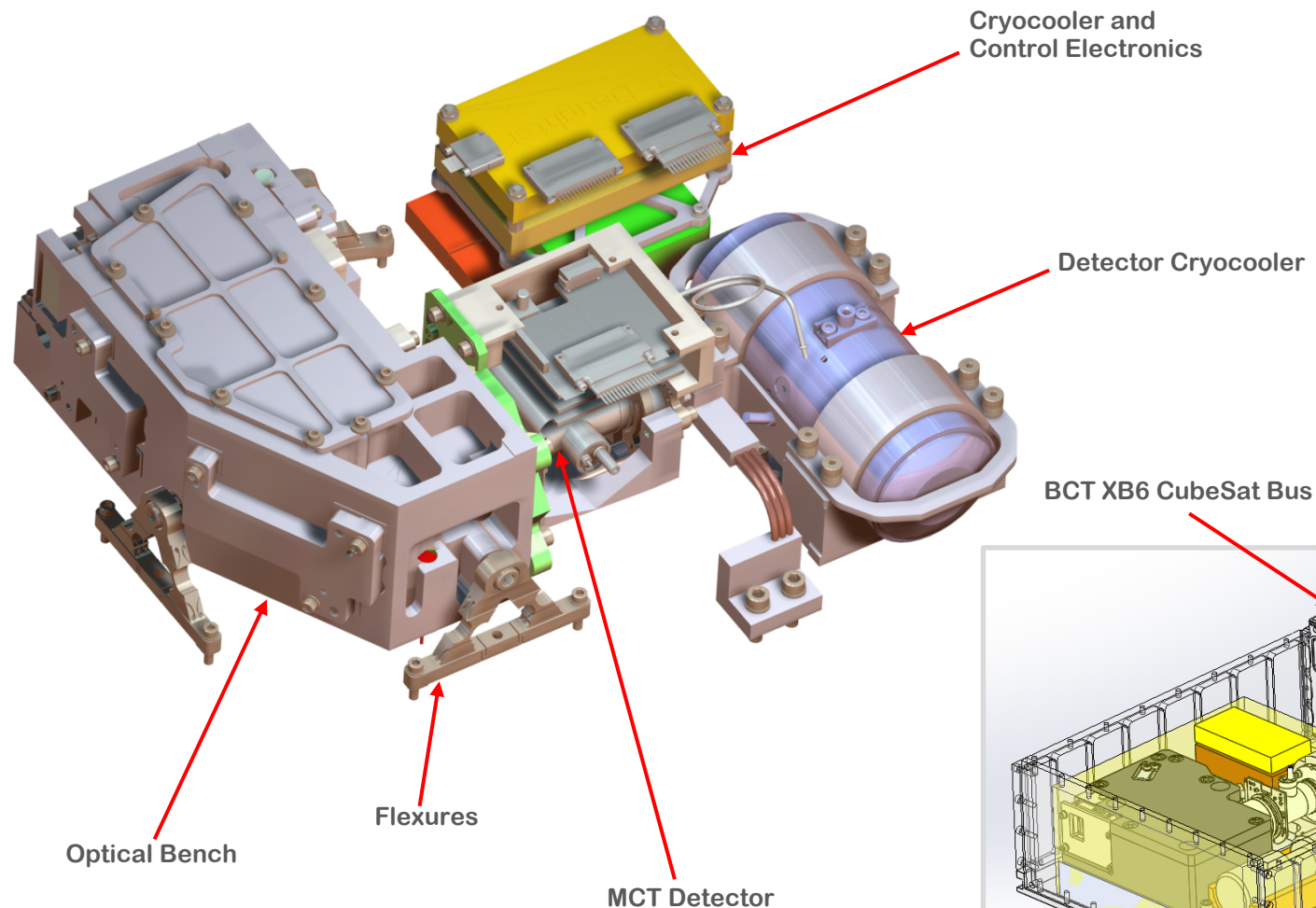
ARCSTONE FSR: All in ONE !



Single channel system, based on new product:
 - Thinned MCT detector
 - Wavelength range from 350 to 2300 nm

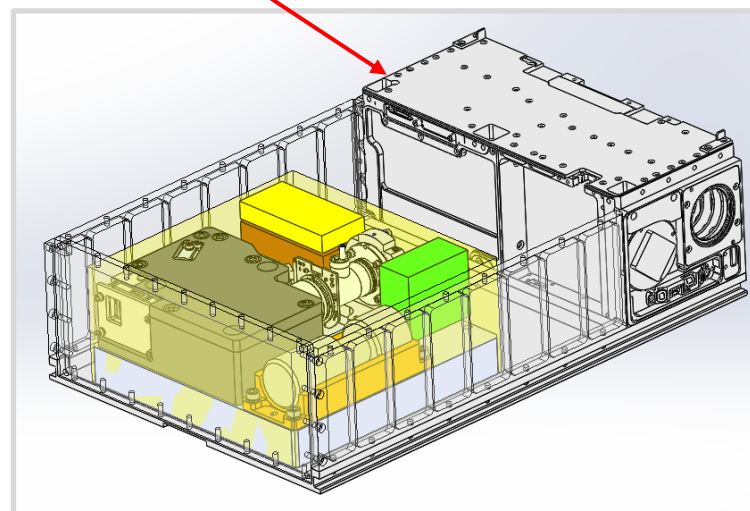
ARCSTONE

Full Spectral Range (FSR) Instrument



Project Approach:

- Achieve TRL5 by March, 2021
- As close to flight as allowed by schedule and budget !



ARCSTONE FSR Status:

- Optomechanical Design
- STOP Analysis in FY19

- LaRC: IIP-QRS-16-0018
- Resonon: SBIR Phase-2



<http://arcstone.larc.nasa.gov>



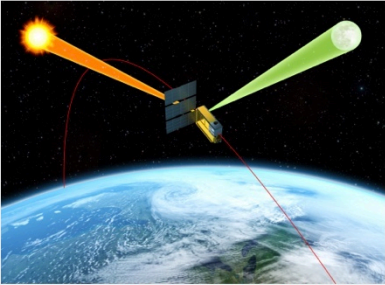
Home About Team Contact



Achieving Instrument High Accuracy In-Orbit

One of the most challenging tasks in remote sensing from space is achieving required instrument calibration accuracy on-orbit. The Moon is considered to be an excellent exoatmospheric calibration source. However, the current accuracy of the Moon as an absolute reference is limited to 5 - 10%, and this level of accuracy is inadequate to meet the challenging objective of Earth Science observations. ARCSTONE is a mission concept that provides a solution to this challenge. An orbiting spectrometer flying on a small satellite in low Earth orbit will provide lunar spectral reflectance with accuracy sufficient to establish an SI-traceable absolute lunar calibration standard for past, current, and future Earth weather and climate sensors.

LEARN MORE



The ARCSTONE observatory is shown in low Earth orbit with the spectrometer viewing the Sun and Moon. The spacecraft rotates in order to view the Moon or the Sun.

“The Moon is available to all Earth-orbiting spacecraft at least once per month, and can be used to tie together the sensor radiance scales of all instruments participating in lunar calibration without requiring near-simultaneous observations.”

– HUGH KIEFFER & TOM STONE

THANK YOU !